

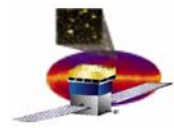
# **GLAST Large Area Telescope:**

## **Electronics, Data Acquisition & Instrument Flight Software**

### **Flight Software – Overview**

**Gunther Haller**  
**Stanford Linear Accelerator Center**  
**Manager, Electronics, DAQ & FSW**  
**LAT Chief Electronics Engineer**

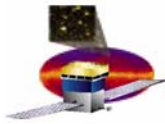
**[haller@slac.stanford.edu](mailto:haller@slac.stanford.edu)**  
**(650) 926-4257**



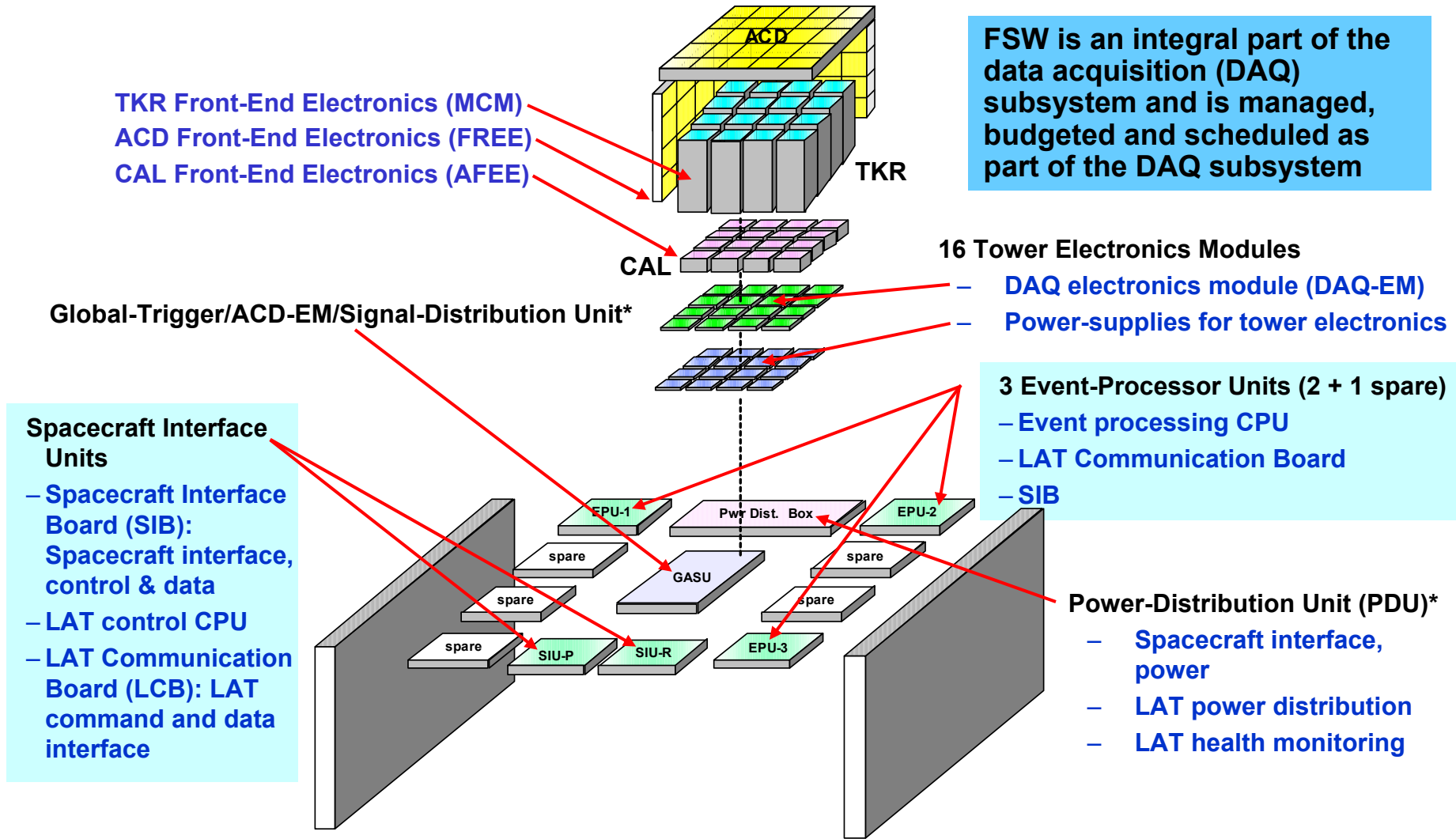
# Outline

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- **DAQ vs. FSW**
- **FSW Overview**
- **Team**
- **Development Plan**
- **Schedule & Milestones**
- **Resource Profile**
- **Documents**
- **Interfaces**
- **Risk Analysis**
- **Changes since PDR**
- **Resource monitoring**



# LAT Electronics



**FSW is an integral part of the data acquisition (DAQ) subsystem and is managed, budgeted and scheduled as part of the DAQ subsystem**

TKR Front-End Electronics (MCM)  
 ACD Front-End Electronics (FREE)  
 CAL Front-End Electronics (AFEE)

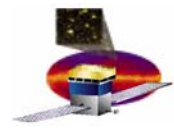
**16 Tower Electronics Modules**  
 - DAQ electronics module (DAQ-EM)  
 - Power-supplies for tower electronics

**3 Event-Processor Units (2 + 1 spare)**  
 - Event processing CPU  
 - LAT Communication Board  
 - SIB

**Spacecraft Interface Units**  
 - Spacecraft Interface Board (SIB):  
 Spacecraft interface, control & data  
 - LAT control CPU  
 - LAT Communication Board (LCB): LAT command and data interface

**Power-Distribution Unit (PDU)\***  
 - Spacecraft interface, power  
 - LAT power distribution  
 - LAT health monitoring

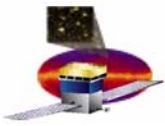
\* Primary & Secondary Units shown in one chassis



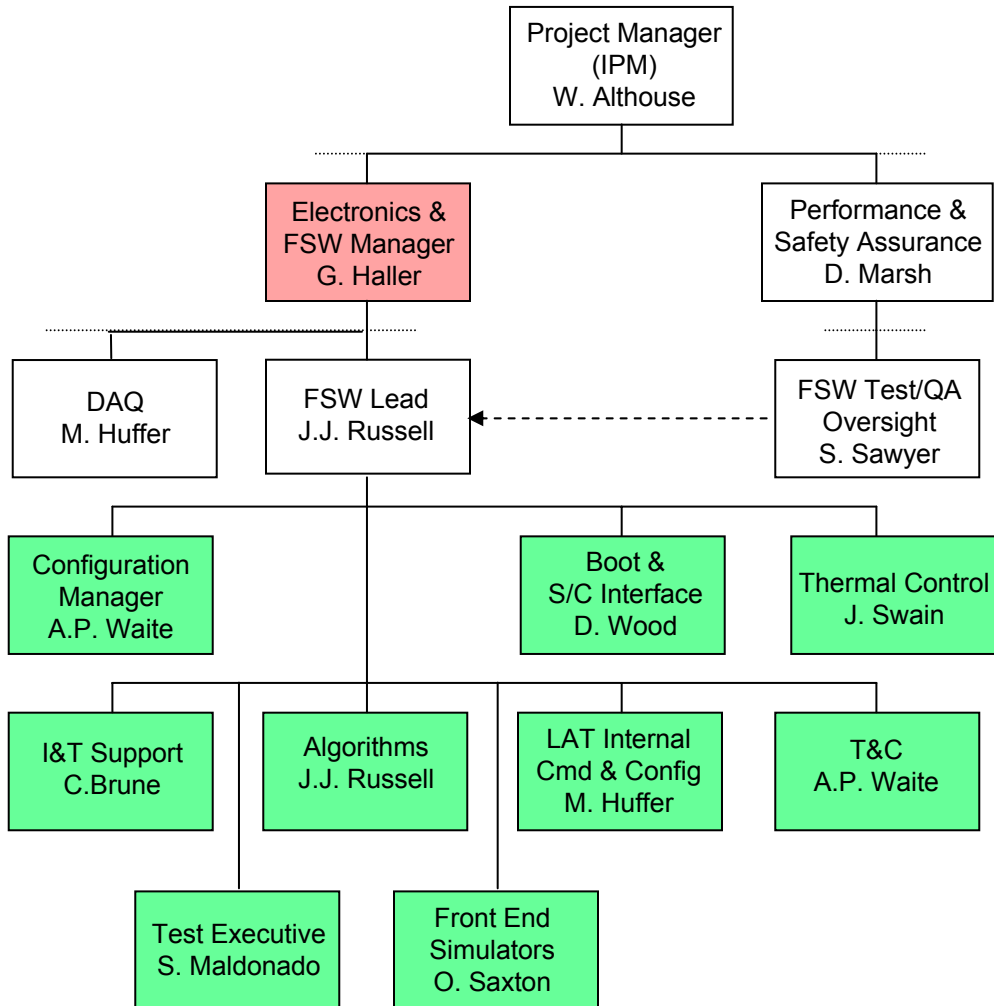
# FSW Overview

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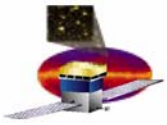
- **LAT FSW is divided into two components**
  - **SIU FSW**
    - **LAT command and control from SC via 1553**
      - LAT hardware configuration and data collection control
      - LAT hardware power control
      - LAT thermal control system
    - **Gathering and distribution of LAT TLM via 1553 and SSR**
    - **Low rate science**
      - Event monitoring for performance information
    - **Transient detection (GRB / AGN)**
      - Alert messages to ground
      - Repoint requests to SC
    - **Calibration / diagnostics**
  - **EPU FSW**
    - **LAT event data processing / filtering**
    - **Calibration / diagnostics**



# Team

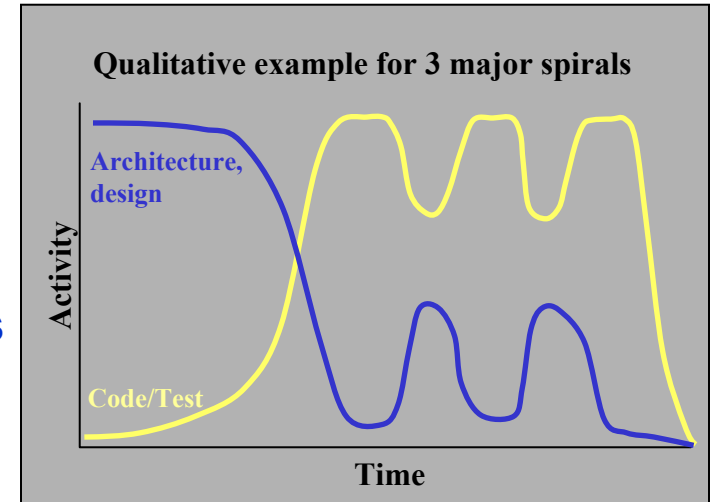


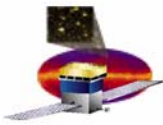
- **Small effective group**
- **Very experienced**
- **Excellent track record**
- **Employ highly interactive development process**
- **All members are expert in LAT architecture, able to contribute in many areas**
- **Leads are highly qualified scientists**
- **Leads are also developers**
- **Independent oversight provided by systems engineering**
- **Produce fully documented design**
- **Process allows/requires software to be in use from early subsystem development/testing to full LAT verification**



# Development Process

- **Initial design effort**
  - Define hardware interfaces and architecture
  - Build stable development infrastructure
  - Generate high-level requirements (SRS) that capture scope of project
  - Generate high-level design that captures basic architecture and interfaces
- **For each FSW release**
  - Generate detailed design of new functionalities
  - Employ iterative design/code/test process to converge on the detailed design (“little spirals”)
    - Allows experienced developer to proceed more rapidly to explore the design parameter space, discover issues, and resolve them
    - Ultimately produces a more optimal design than one selected in advance based only on analysis and limited data
    - Extensive documentation of resulting code is produced as it is built
    - Iterative process is a continuous rapid prototyping cycle that supports higher productivity and a higher quality final product





# Code Management

- **FSW partitioned into functional blocks based on the SRS**
  - **Functional blocks are then mapped into packages, the fundamental unit of the code management system**

## Common code – SIU and EPU

PACKAGE	DESCRIPTION	LOC	RISK FACTOR	CONTINGENCY
PBS	Basic Services	1800	1,2	75
FMP	File Management	1500	3	450
CHP	CPU Housekeeping	1000	3	300
SWD	Software Watchdog	1000	3	300
LIO	LAT LCB I/O driver	3000	5	1500
HUT	Histogrammer	1750	3	525
SSR	SSR services	1000	5	500
SDF	SIU Diag Framework	1000	10	1000
<b>TOTALS</b>		<b>12050</b>		<b>4650</b>

## EPU specific code

PACKAGE	DESCRIPTION	LOC	RISK FACTOR	CONTINGENCY
EDP	Event Dispatch	2100	3	700
EOP	Event Output	3750	3	1125
EMP	Event Monitoring	3500	2	700
ECL	Calibration (client)	4950	5	2475
ECP	Command/Control	3400	3	1020
SOP	Event output	1250	2	250
EFP	Event Filtering	10000	5	1000
<b>TOTALS</b>		<b>28950</b>		<b>7270</b>

## Test and verification code

PACKAGE	DESCRIPTION	LOC	RISK FACTOR	CONTINGENCY
FES	Front end simulator	5600	3	1680
LTX	Test executive	5200	5	2400
<b>TOTALS</b>		<b>10800</b>		<b>4080</b>

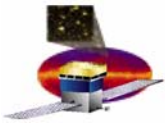
\*See next slide for discussion of contingency

## SIU specific code

PACKAGE	DESCRIPTION	LOC	RISK FACTOR	CONTINGENCY
GNAT	LAT DAQ cmd/resp	1550	1	155
GCFG	DAQ configuration	5850	1,3	925
LCP	LAT commanding/1553	3600	2	360
SMP	Event output	3100	3	930
SCL	Calibration (server)	3200	5	1600
HSK	Housekeeping	6350	3	1905
THP	Trend/bounds check	1500	2	300
MCP	Mode control	1300	3	390
GPS	GPS services	800	3	240
LAP	SC attitude services	1300	3	390
GBM	Response to GBM data	1550	3	465
GRB	GRB detection	2400	3	720
TCS	Thermal Control	2000	5	1000
<b>TOTALS</b>		<b>34500</b>		<b>9380</b>

## Boot code

PACKAGE	DESCRIPTION	LOC	RISK FACTOR	CONTINGENCY
REV	Reset/exceptions	400	2	40
MEM	Memory test	100	1	10
BSH	Boot shell	1000	3	300
EDV	EEPROM driver	100	1	5
PCI	PCI driver	300	0	0
CCSDS	CCSDS formatter	400	0	0
UPL	Upload parser	1000	3	300
ELF	ELF dynamic loader	500	0	0
ZLIB	ZLIB integration	40	0	0
BUT	Boot Utilities	200	0	0
LCB	LCB driver (polled mode)	400	5	200
CO1553	1553 driver (polled mode)	400	0	0
<b>TOTALS</b>		<b>4840</b>		<b>855</b>

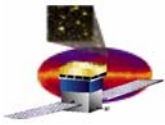


# Contingency

- Document: LAT-TD-01781
- LOC count estimated for each package
  - Bottoms up estimate
  - Based on package content
  - Based on previous experience
- LOC estimates not worst-case
  - Represent most likely length of package
  - Some longer or more complex, some shorter or less complex
  - Total LOC for entire FSW load expected to be more static than individual package estimates

RISK FACTOR	CRITERIA
0	Written
1	Significant coding already complete, also testing or previous experience (e.g. balloon flight)
2	Partial coding already complete, also limited or very well defined scope
3	Well understood scope, possibly some coding completed
5	Scope partially undefined, limited previous experience or high complexity
10	Scope poorly defined

- Contingency estimation based on risk factor
  - Risk factor defined in table above
  - Contingency calculated by multiplying the risk factor by 10% of total estimated LOC in package
  - Contingency represents potential additional lines of code

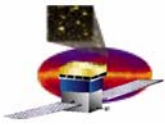


# Releases

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- **LAT FSW strategy calls for major FSW releases to coincide with the natural hardware builds as follows:**
  - **R1 – Engineering Model 1 (8/1/03)**
    - **Single tower, single CPU**
  - **R2 – Engineering Model 2 (1/1/04)**
    - **Multiple towers (single tower plus front-end simulators for additional towers), GASU\*, single CPU**
  - **R3 – Full LAT (9/1/04)**
    - **Complete set of 16 towers, GASU\*, full set of CPU's**

\* GASU includes LAT Global Trigger (GEM), ACD Electronics Module (AEM), LAT Command Response Unit (CRU) and Event-Builder (EB)



# EM1 Release

## • Hardware

- 1 Partially populated tower
- 1 Tower Electronics Module
- 1 COTS CPU (VME)
  - Ethernet
  - Serial port
  - LCB communications
    - Command/response
    - Event acquisition

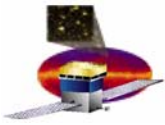
## • Software

- Interfaces (other than VxWorks)
  - LCB command/response
  - LCB event acquisition
- TEM configuration setting and read-back
  - Write to and read from all tower registers
  - CAL, TKR, TEM
- Format and export event data from tower
- Charge injection calibration
  - Inject a known charge signal directly into the (TKR, CAL) electronics in lieu of the detector output
  - Read the resulting event data output
- Sample and collect a subset of tower / TEM housekeeping and LRS data
- Continue EPU filter development and testing separately on desktop machine

### Status:

Development complete against preproduction electronics

Deployed to field in I&T test stands



# EM2 Release

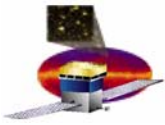
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- **Hardware**

- Multiple towers (single tower plus FESs)
- Multiple TEMs
- GASU (or simulation)
  - Command Response Unit (CRU)
  - Event Builder Module (EBM)
  - ACD Electronics Module (AEM)
  - Global Trigger Electronics Module (GEM)
- 1 COTS “SIU/EPU” CPU (cPCI)
  - Ethernet
  - Serial port
  - SIB
  - LCB

- **Software**

- All of EM1 functionality
- Multiple tower capabilities
- AEM configuration
- AEM event acquisition
- Capability to inject marker events into event streams to provide notice of filter parameter changes
- LAT mode transitions
  - Engineering and safe modes
- LAT spacecraft interface
  - 1553
  - Command and telemetry
- File management system
- “Desktop” EPU for filter testing and troubleshooting



# Full LAT FSW Release

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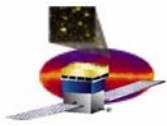
- **Hardware**

- All towers / TEMs
- ACD
- GASU
  - Command Response Unit (CRU)
  - Event Builder Module (EBM)
  - ACD Electronics Module (AEM)
  - Global Trigger Electronics Module (GEM)
- Development on multiple engineering RAD750s to emulate flight SIU/EPUs

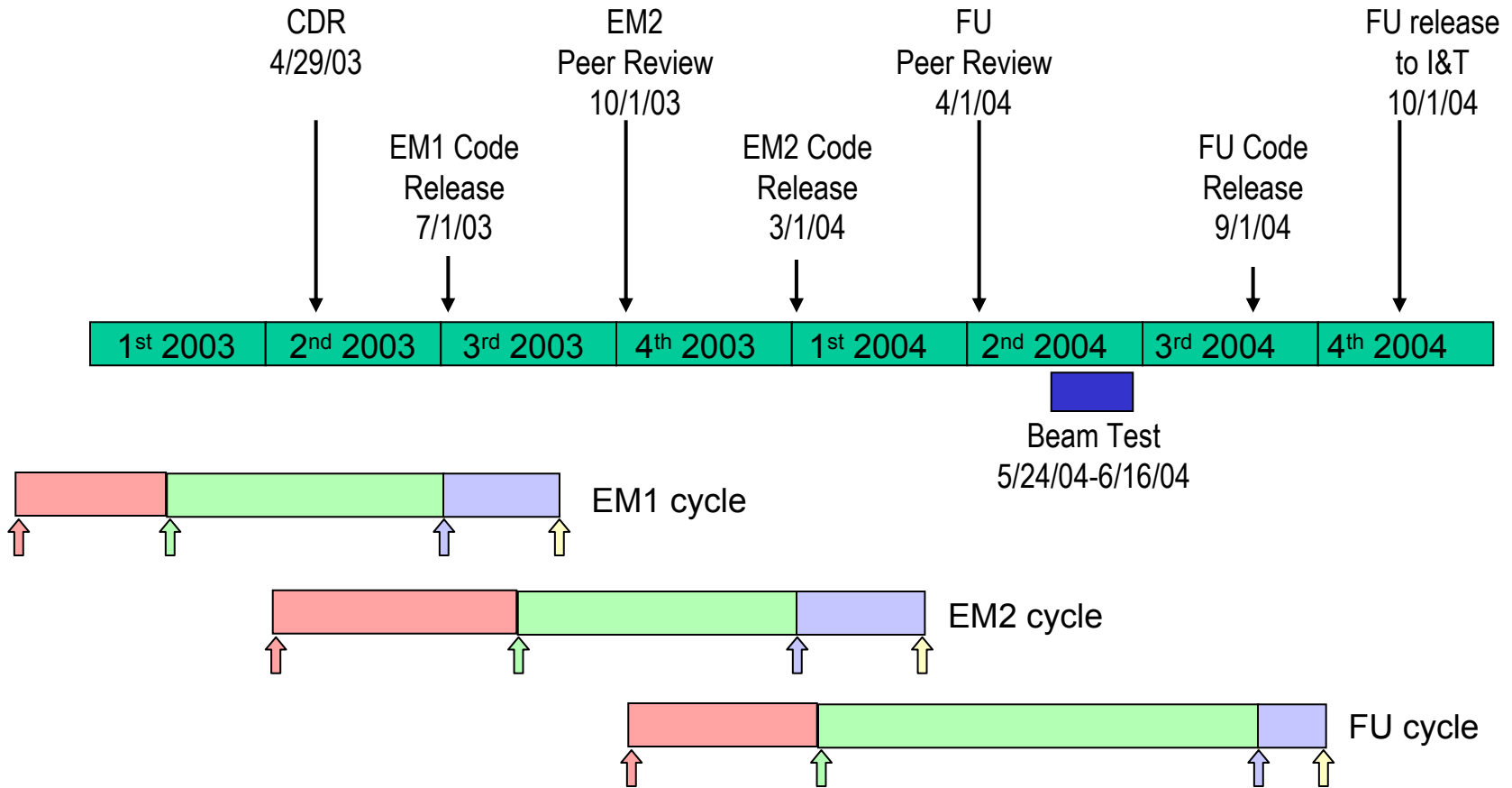
- **Software**

- All of EM2 functionality
- Multiple processor capabilities
  - EPU configuration by SIU
- Spacecraft message processing
  - Attitude, time, ancillary data
- Transient detection and reporting
- LAT hardware power control
- Thermal control system





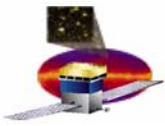
# Breakdown of Development Cycles





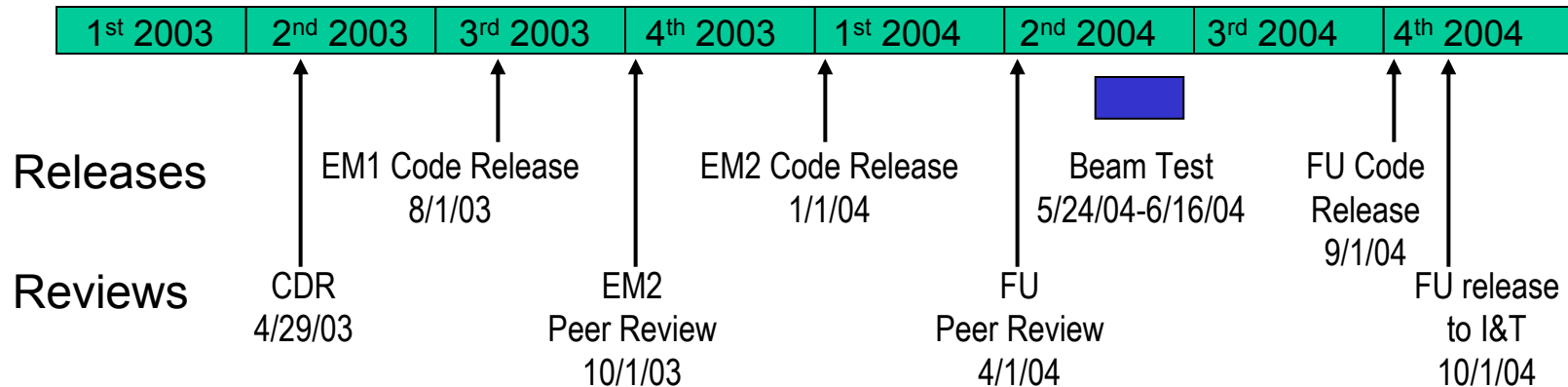




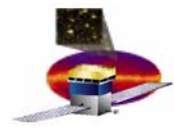


# Milestones

- Detailed FSW development plan, schedule, and reviews are part of LAT PCMS, down to the task level (WBS 4.1.7.9)
- Plan identifies milestones for progress assessment



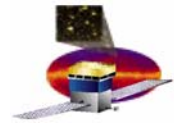
- LAT CDR will serve as review of high-level FSW design and detailed EM1 design
- Incremental milestones for each package completion
  - Detailed in FSW Development Schedule
  - Will additionally be tracked and discussed in
    - Weekly LAT-wide project meetings with discussion of each sub-system
    - LAT Project Weekly Report
    - Monthly LAT-wide PMCS reviews: system-specific past month's accomplishments, plans for following month, risk evaluation, cost and schedule review of last month's scheduled and budgeted versus actual accomplishments



# Resource Profile

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- **Personnel resources**
  - **SLAC: 6 FTE**
  - **NRL: 2 FTE (Dan Wood plus two additions)**
    - **Brian Davis (20%): extensive experience with software requirements and engineering as well as code development**
    - **Ray Caperoon (80%): coming to us from SECCHI with a background in RAD750 programming for that program**
- **Period of performance**
  - **19 months currently shown until LAT Pre-Ship Review**
  - **Assume 18 months available for FSW development**
- **Overhead**
  - **Schedule assumes**
    - **LAT FSW Team Leads – 20% maximum**
    - **LAT FSW Team (SLAC) – 10% maximum**
    - **LAT FSW Team (NRL) – 15% maximum**
    - **Above figures reflect levels from previous experiments**

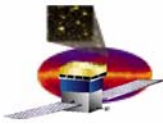


# Cost by Fiscal Year

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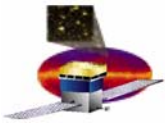
- **Flight Software without contingency**

<b>FY01 in K\$</b>	<b>FY02 in K\$</b>	<b>FY03 in K\$</b>	<b>FY04 in K\$</b>	<b>FY05 in K\$</b>	<b>Total in K\$</b>
<b>70</b>	<b>223</b>	<b>935</b>	<b>713</b>	<b>743</b>	<b>2,684</b>



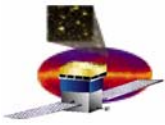
# FSW Documentation

Document	Title	Version	Status	Comment
<i>Specifications</i>				
433-SPEC-0001	Level 2(a): Mission System Specification			
LAT-SS-00010	Level 2(b): LAT Instrument Performance Specification	2	CM	
LAT-SS-00019	Level 3: T&DF Subsystem Specification	2	CM	
LAT-SS-00284	Level 4: Trigger Specifications	0	CM	
LAT-SS-00285	Level 4: Dataflow Specifications	0	CM	
LAT-SS-00399	Level 4: Software Requirements Specifications (SRS)	1	CM	Release prior to CDR
<i>Introductory Documents</i>				
LAT-SS-00287	Conceptual Design of the Electronics, T&DF System	0	PDR	
LAT-TD-00458	GLAST LAT Electronics, DAQ, Flight Software Preliminary Design Report	1	PDR	
LAT-TD-00518	Preliminary Design Report (See 458)	1	PDR	
LAT-TD-331	LAT Flight Software Preliminary Design Report	0	PDR	
<i>External Interface Documents</i>				
GSFC-433-IRD	LAT-SC Interface Control Document			
LAT-SS-00363	ACD-DAQ Interface Specifications	4	CM	
LAT-SS-00238	CAL-DAQ Interface Specifications	4	CM	
LAT-SS-00176	TKR-DAQ Interface Specifications	1	CM	
LAT-SS-00467	CAL-AFEE-TEM Interface Control Document	1	CM	
<i>Software ICDs</i>				
LAT-TD-00860	LAT Communication Board (LCB) Programming ICD	2	100%	Release prior to CDR
LAT-SS-01543	Power Distribution Module Programming ICD Specifications	0	50%	Release prior to CDR
LAT-TD-00639	ACD Electronics Module Programming ICD Specifications	1	90%	Release prior to CDR
LAT-TD-00605	Tower Electronics Module - A Primer ICD Specification	1	100%	Release prior to CDR
LAT-TD-01545	Global Trigger Programming ICD Specification	0	100%	Release prior to CDR
LAT-TD-01546	Event Builder Programming ICD Specification	0	90%	Release prior to CDR
LAT-TD-01547	Command Response Unit Programming ICD Specification	0	90%	Release prior to CDR



# FSW Documentation (cont.)

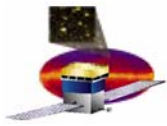
Document	Title	Version	Status	Comment
<i>ICDs</i>				
LAT-SS-01539	LAT Spacecraft Interface Board (SIB) ICD Specifications	0	100%	Release prior to CDR
LAT-SS-01538	LAT Crate Power Supply ICD Specifications	0	100%	Release prior to CDR
LAT-SS-01540	LAT Crate Backplane ICD Specifications	0	100%	Release prior to CDR
LAT-SS-00169	Conceptual Design and Specification of the GLAST Tracker Front-End Electronics (GTFE) ASIC	4	CM	
LAT-SS-00170	Conceptual Design of the GLAST Tracker Readout Controller Electronics (GTRC) ASIC	0	CM	
LAT-TD-01550	Global Tracker Cable Controller (GTCC) ASIC ICD	0	10%	Release prior to CDR
LAT-DS-00087	Conceptual Design of the GLAST Calorimeter Front-End Electronics (GCFE) ASIC	3	CM	
LAT-SS-00208	Conceptual Design of the GLAST Calorimeter Readout Control (GTRC) ASIC	0	CM	
LAT-TD-01549	Global Calorimeter Cable Controller (GCCC) ASIC ICD	1	50%	Release prior to CDR
LAT-TD-01548	Global Trigger Integrated Circuit (GTIC) ASIC ICD	2	Draft	Release prior to CDR
<i>Support Equipment Documents</i>				
LAT-TD-00861	EGSE Test-stand Architecture Redux	1	100%	Release prior to CDR
LAT-TD-00862	EGSE Test-stand Update	1	100%	Release prior to CDR
LAT-TD-00593	EGSE LAT COMM IO Board Response FIFO	4	Draft	Release prior to CDR
LAT-TD-01824	Front-End Data Simulator Description		40%	Release prior to CDR
LAT-TD-01825	Front-End Data Simulator Programming ICD		40%	Release prior to CDR
<i>Plans and Management Documents</i>				
LAT-MD-00104	Flight Software Management Plan	0	CM	Release prior to CDR
LAT-TD-00786	Flight Software Test Plan	0	CM	Release prior to CDR
LAT-PR-01535	Flight Software Development Plan	1	100%	Release prior to CDR
LAT-TD-01826	DAQ/FSW Test/Verification Plan			



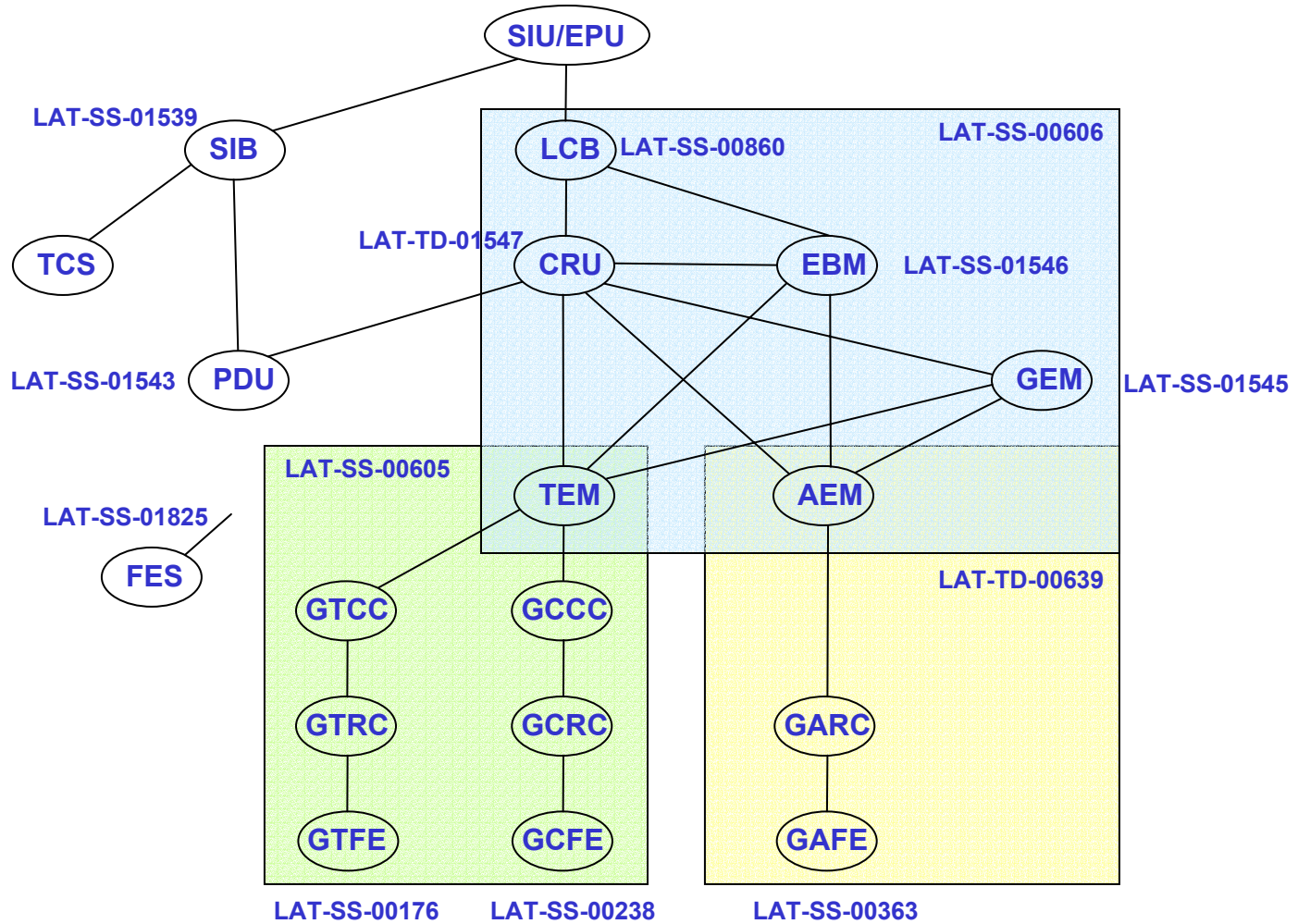
# FSW Documentation (cont.)

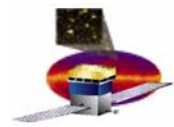
Document	Title	Version	Status	Comment
<i>Other Relevant Documents</i>				
LAT-SS-00286	Global Trigger Description	0	100%	Release prior to CDR
LAT-TD-00606	LAT Intermodule Communication	1	100%	Release prior to CDR
LAT-TD-01380	LAT Communication Board Driver - SW Architecture and Interfaces	2	Draft	Release prior to CDR
LAT-SS-00461	LAT TEM-GASU to CPU Data Formats	1	100%	Release prior to CDR
LAT-TD-01536	LAT ICD for Start-up Procedures	0	100%	Release prior to CDR
LAT-SS-00890	LAT Instrumentation ICD		100%	Release prior to CDR
LAT-TD-00712	LAT Auxiliary Data Survey	0	Draft	Release prior to CDR
LAT-SS-00715	Thermal Control System ICD	0	Draft	Release prior to CDR
LAT-TD-01553	Instrument Damage Protection against LAT Processor Hardware or Software Malfunction	0	90%	Release prior to CDR
LAT-TD-00863	LAT Custom Processor Specifications	1	Draft	
LAT-TD-01781	LAT Flight Software Package Descriptions and LOC Basis of Estimate	0	100%	
	CMX Manual	0	Draft	
LAT-TD-01806	SIU Boot Code	0	90%	
	CCSDS Packet Library User Manual*	0	Draft	
	SIU 1553 Drivers*	0	Draft	
	Telecommand and Telemetry Formats*	0	Draft	
LAT-TD-01121	LAT DAQ Trigger and Dataflow Resource Usage		100%	
LAT-SS-01597	Configuration Data: Storage and Transmission	0	90%	
LAT-SS-00xxx	Software/Algorithm Design Document			EM1 Release

\* Documents produced at NRL and not yet entered into CyberDocs



# FSW ICD Tree

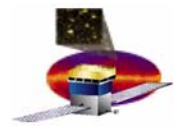




# Risk

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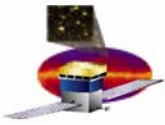
- **LAT instrument FSW not critical to mission safety**
  - **No LAT commands or FSW actions can result in damage to hardware**
  - **All relevant hardware has built-in self-protection**
    - **Current limiting protects PMTs during SAA if HV reduction not performed in advance by ground command or FSW action**
- **LAT instrument FSW supports mission success**
  - **Extensive ground testing (as with balloon flight) prior to flight**
  - **Use of FSW for electronics and system level testing to achieve more user hours by non-developers**
  - **Reprogrammability**
    - **Fully reprogrammable on orbit except for small amount of primary boot code**
    - **Primary boot code being treated as critical code**



# Changes Since PDR

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- **Processor selection**
  - **BAE RAD750 has become baseline processor**
  - **Number of processors has been determined**
    - **2 SIU (1 cold spare)**
    - **3 EPUs (2 active, 1 cold spare)**
  - **SIU and EPU crates now look alike**
- **Event Builder has become part of GASU**
  - **Some SIU code has migrated to EPU or common code**
  - **SSR is now a node on event fabric**



# Resource Usage Monitoring

Resource	Available	Used	Margin Factor
SIU Boot Prom	256 KBytes	128 KBytes	2
EPU Boot Prom	256 KBytes	128 KBytes	2
SIU EE Prom	4 MBytes	1.5 - 2.5 MBytes	1.5 - 2.5
EPU EE Prom	4 MBytes	1 - 2 MBytes	2-4
SIU DRAM Code/Data	128 MBytes	4-8 / 32-96 <sup>1</sup> MBytes	1.25 - 3
EPU DRAM Code/Data	128 MBytes	4-6 / 32-96 <sup>1</sup> MBytes	1.25 - 3
SIU CPU Cycles	100% in 1 CPU	25%	4
EPU CPU Cycles	200% in 2 CPUs	30 - 50%	4 - 6
XEM-to-CPU Bandwidth	20.0 MBytes/sec	10.0 MBytes/sec <sup>2</sup>	2
CPU-to-CPU Bandwidth	2.5 MBytes/sec	50.0 KBytes/sec <sup>2</sup>	50
CPU-to-SSR Bandwidth	5.0 MBytes/sec	40.0 KBytes/sec <sup>2</sup>	125

<sup>1</sup> Data memory usage is largely a function of how much monitoring data is kept. This is a soft requirement that can be adjusted before and during flight.

<sup>2</sup> Physics Data Taking Mode

Table assumes 1 active SIU (1 cold spare), 2 active EPUs (1 cold spare)

All CPUs BAE RAD750s (133 MHz, 128 MB memory, 256 kB on board PROM for primary boot)