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Document Title LAT FSW Qualification Test Procedure: EVTFIL 002: Event Filtering – Event Filter Rates and Capacity		

CHANGE HISTORY LOG

Revision	Effective Date	Description of Changes
1	08/15/05	Original
2	01/04/06	Updates for FSW B0-6-1 interfaces

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1.SCOPE

This document describes the procedure followed for qualification testing of the LAT FSW. This Qualification Test Procedure document describes one of the qualification tests executed to verify compliance with the requirements defined in the “Flight Software Specification – Level III” (LAT-SS-00399).

1.1 Test Suite

EVTFIL

The Test Suite is a logical grouping of a set of tests based on common functionalities and can be executed as a suite of tests in some order when the pre-conditions and post-conditions for each test within the suite have been met successfully.

This test procedure *EVTFIL_002* belongs in the Test Suite *EVTFIL* under the *FST* Project. The *EVTFIL* tests verify the functionality of the LAT onboard event filter software.

1.2 Test ID

EVTFIL_002

Test *EVTFIL_002* contains one main test script, *evtfil_002.py*, which can be run when the SIU and EPU are operating in Boot Mode.

1.3 Requirement(s) Tested

The Qualification Test Procedure described herein is performed to verify that the FSW satisfies the following requirement(s), quoted from the Flight Software Specification – Level III:

Requirement Number	Requirement Name	Requirement
5.3.9.1	Event Filter Rates and Capacity	<p>The FSW shall provide the capability to filter the input stream of events accepted by the electronic trigger to an output stream commensurate with the spacecraft (SC) interface rate (see 5.2.1.3.1), and capacity of a maximum of 104 Gb in any given 24 hour period, keeping events meeting the science objectives.</p> <p>Note that the 104 Gb allocation includes a total source packet overhead of 112 bits per packet (64 bits of the secondary header are for a time stamp).</p>

If the requirement(s) quoted above cite external documents (e.g., “...Further details are provided in [11]”), consult LAT-SS-00399 for the list of citations.

2.DEFINITIONS AND ACRONYMS

The following terms, abbreviations, and acronyms are used in this document:

2.1Definitions

Hz	Hertz, unit of frequency
s, sec	seconds
V	Volt
W	Watt
s, sec	seconds

2.2Acronyms

CAL	Calorimeter
EGSE	Electrical Ground Support Equipment
FOV	Field of View
GASU	Global trigger Anti-collision Spacecraft Unit
GBM	Gamma-ray Burst Monitor
GRB	Gamma Ray Bust
PTR	Post Test Review
TEM	Tower Electronics Module
TKR	Tracker
TPS	Tower Power Supply
TRR	Test Readiness Review
QAE	Quality Assurance Engineer
TE	Test Engineer
BPU	Bus Protection Unit
APID	Application Identifier
FES	Front-End Simulator
VSC	Virtual Spacecraft
LIM	LAT Instrument Manager
LPA	LAT Physics Application
SIU	Spacecraft Interface Unit

3.REFERENCES

The list below provides documents that are to be used as references for this procedure:

3.1Applicable Documents

<u>Document Number</u>	<u>Description</u>
<u>SPECIFICATIONS</u>	
LAT-SS-00399	LAT Flight Software Level III Specification
1196 EI-S46310-000	GLAST 1553 Bus Protocol Interface Control
LAT-TD-02659	LAT Flight Software Telecommand and Telemetry Formats
LAT-TD-0561	The Virtual Spacecraft (VSC)
LAT-SS-00010-1	GLAST LAT Performance Specification - Level II(b) Specification
<u>PROCEDURES</u>	
<u>PLANS</u>	
LAT-MD-00039	Performance Assurance Implementation Plan
LAT-MD-00078	GLAST LAT System Safety Program Plan
LAT-MD-00404	LAT Contamination Control Plan
LAT-MD-00408	LAT Program Instrument Performance Verification Plan
LAT-SS-00296	T & DF Test Plan
LAT-TD-00297	LAT Electronics Test Plan
LAT-TD-00786	LAT Flight Software Test Plan
LAT-MD-00104	LAT Flight Software Management Plan
<u>DRAWINGS</u>	
<u>OTHER</u>	
LAT-MD-00091	GLAST Quality Manual
LAT-MD-00471	Control of Nonconforming Product
LAT-MD-00472	Corrective and Preventative Action

4.REQUIREMENTS

This section lists the requirements that shall be followed during the LAT FSW Qualification Testing process.

The Performance Assurance Implementation Plan, LAT-MD-00039, shall be utilized to ensure that the products produced by the GLAST LAT project intended for design qualification, flight and critical ground support equipment usage meet the required levels of quality and functionality for their intended purposes.

The LAT Program Instrument Performance Verification Plan, LAT-MD-00408, shall be utilized to address the testing to be performed at the unit/subsystem and instrument level for flight qualification, proto-flight and acceptance testing phases. Also included are the necessary processes/procedures and systems assurance activities.

4.1 Test Data, Equipment and Software

This procedure shall follow the requirements found in the Control of Nonconforming Product, LAT-MD-00471. This document establishes the method to identify and control nonconforming product developed by the LAT project team.

4.2 Quality Assurance

This procedure shall follow the requirements found in the Corrective and Preventative Action document, LAT-MD-00472 and the GLAST Quality Manual, LAT-MD-00091.

The Corrective and Preventative Action document establishes the method to be used to initiate, implement, evaluate and record corrective and preventive actions. The GLAST Quality Manual defines the methods implemented by the GLAST LAT project to ensure consistent quality of all processes for procurement, design, development and production of flight hardware, flight software and all associated ground support equipment interfacing with flight hardware and software.

4.3 Safety

This procedure shall follow the requirements found in the GLAST LAT System Safety Program Plan, LAT-MD-00078. This document defines all phases of the LAT program including: design, development, fabrication, handling, transportation, storage, test, assembly and operation.

When high voltages are present extreme care should be exercised.

4.4 Warnings, Cautions, and Notes

The following SAFETY ALERTS are intended to create awareness of the potential safety hazards and the steps that must be taken to avoid accidents. These same alerts are used throughout this document to identify specific hazards that may endanger personnel and/or equipment.

Identification of every conceivable hazardous situation is impossible. Therefore, all personnel have the responsibility to diligently exercise safe practices whenever exposed to this equipment.

Indicates a potential hazardous situation which, if not avoided, could result in death or injury.

Indicates a potential hazardous situation which, if not avoided, could result in damage to equipment.

Indicates a notification of information that is important, but not hazard related.

4.5 General Instructions

This qualification test procedure shall be conducted on a formal basis to its latest approved and released version. The designated Software QAE shall be notified 24 hours prior to the start of this procedure. Software QAE may monitor the execution of all or part of this procedure should they elect to do so.

The Test Engineer conducting this test shall read this document in its entirety and resolve any apparent ambiguities before beginning the procedures described herein.

Deviations from the procedures described in this document and breaks in hardware or software configuration can only be initiated by the Test Engineer, must be approved by QA, and must be documented in Appendix A.

Any nonconformance/defect/anomaly is to be reported in JIRA. Refer to the LAT Flight Software Test Plan LAT-TD-00786 for guidance. Do not alter or break configuration if a failure occurs. Notify Software Quality Assurance.

All success conditions for a test must be met for the test to pass.

5.SETUP

This section describes the hardware and software configuration used for the qualification test described later in this document. Any break from configuration or deviation from a particular procedure must be authorized by the Quality Assurance Engineer and documented in Appendix A.

5.1Hardware Setup

The list below indicates the equipment that is used to execute the tests described in this document.

Hardware Unit	Manufacturer	GLAT Number	Firmware Version (where applicable)
Front End Simulator (FES): 1	SLAC		
Virtual Spacecraft (VSC): 1	SLAC		
Spacecraft Interface Unit (SIU): 1	SLAC		
Event Processing Unit (EPU): 2	SLAC		
Tower Electronics Module (TEM): 16	SLAC		
Global Trigger AEM Signal distribution unit (GASU): 1	SLAC		
Power Distribution Unit (PDU):1	SLAC		
Unix or Linux host Establishes connection between VSC and the terminal from which the test is run.			

The Figure below depicts the Testbed on which this qualification test is performed. The particular hardware units utilized in this test are itemized by GLAT number and firmware version in the preceding table.

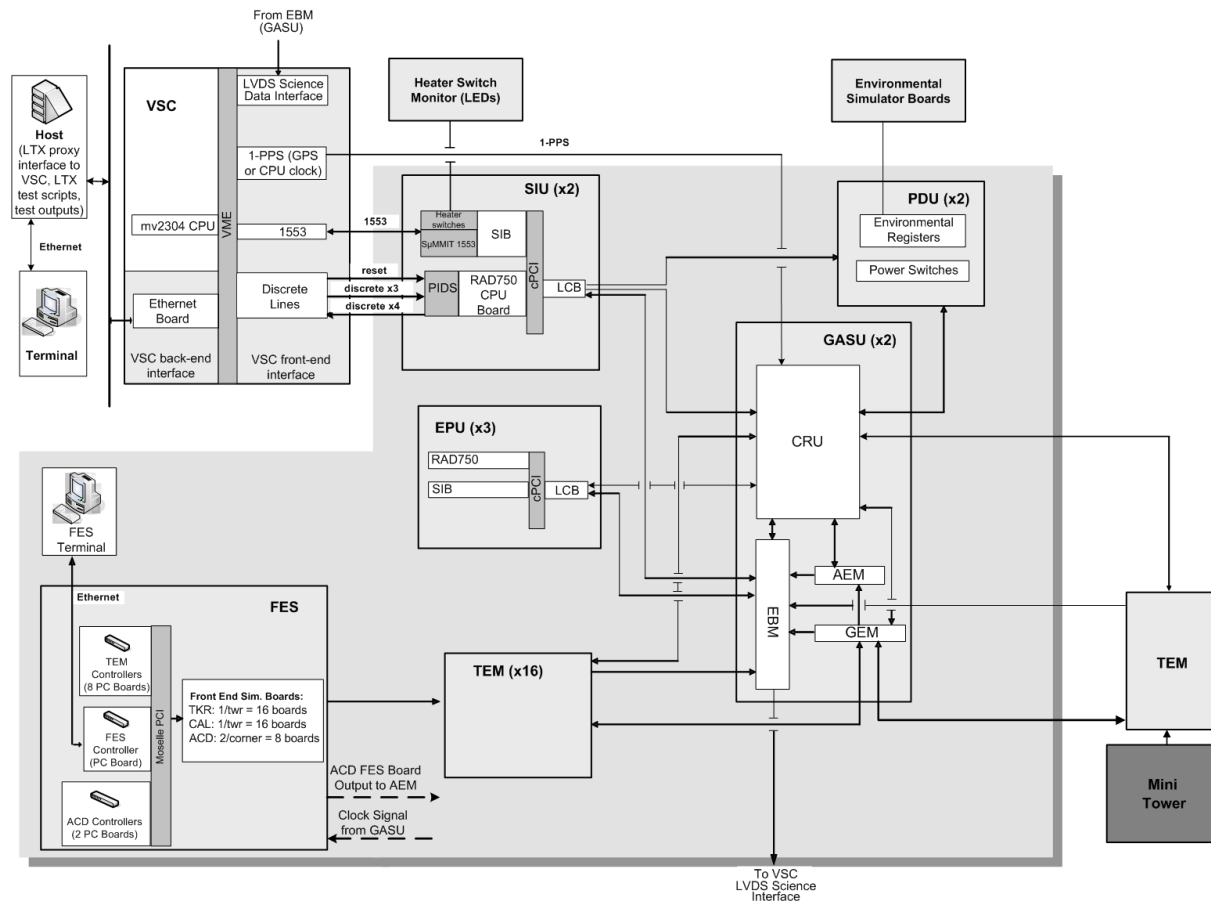


Figure 1. FSW Testbed

5.2 Software Setup

The software required to prepare for and execute the tests described in this Qualification Test Procedure document is itemized in this section.

5.2.1 Test Tools

The following table specifies the test executive used to run this qualification test, and identifies the other software tools used to support the execution of the test. The “Software Version Number” column identifies the version number of the test tool being used; alternatively, this column identifies the hardcopy attachment to this document that records the version of the tool being used (e.g., “Attachment 1”). The “Path to Attachment” column identifies the directory in which the electronic copy of any hardcopy attachment is saved (if applicable).

Software	Description of Software	Software Version Number (or Specify Attachment Number)	Path to Attachment (If Applicable)
LTX	LAT Test Executive		

Software	Description of Software	Software Version Number (or Specify Attachment Number)	Path to Attachment (If Applicable)
LCAT	Telecommand creation and telemetry decoding software		
FMX	File Management Extra		
VPI	VSC Proxy Interface		
VSCP	Python API for Virtual Spacecraft software		
STU	Standard Test Utilities		

5.2.2 Test Scripts

The following table identifies the test scripts that are run to execute this qualification test. The “Script Version Number” column identifies which version of the script is being used; alternatively, this column identifies the hardcopy attachment to this document that records the version of the script being used (e.g., “Attachment 1”). The “Path to Attachment” column identifies the directory in which the electronic copy of any hardcopy attachment is saved (if applicable).

Test Script	Description of Test Script	Script Version Number (or Specify Attachment Number)	Path to Attachment (If Applicable)
evtfil_002.py	Main LTX qualification test execution script for EVTFIL_002		
EVTFIL_002.xml	LTX test description file for EVTFIL_002		
evtfil_002_analyze.py	Test data/results analysis script for EVTFIL_002		

5.2.3 Flight Software

This qualification test is performed on a complete, integrated Candidate Release of FSW. All FSW libraries under test are final Flight Unit Candidate versions. The test described in this document is designed to evaluate the particular FSW packages and constituents listed in the following table; for clarity, only those constituents that are the focus of the test are listed below.

FSW Package	Constituent(s)
EFC	

The Candidate Release into which these FSW constituents are integrated is identified in the following table. The “Candidate Release Build Number” column specifies the build number of the FSW Candidate Release under test (e.g., “B0-0-1”). The “Candidate Release Attachment Number” column identifies which hardcopy attachment to this document confirms the name and version number of the full set of FSW packages and constituents used in the test (e.g., “Attachment 2”). The “Path to Attachment” column specifies where in the test repository an electronic copy of the hardcopy attachment has been saved.

Candidate Release Build Number	Candidate Release Attachment Number	Path to Attachment

5.3 Setup Validation

5.3.1 Hardware Validation

The following signatures confirm that the Test Engineer and Quality Assurance Engineer have verified the GLAT numbers, firmware version numbers, and the proper connection of all hardware listed in the table in Section 5.1.

 Date Time Test Engineer QAE

5.3.2 Software Validation

The Test Engineer performs the following procedure to validate the software setup for this qualification test and records completion of the setup validation steps in the space provided.

Step No.	Description of Step	Step Outcome
1	Record the version numbers of all test tools used to perform this qualification test in the table in Section 5.2.1.	Complete/ Not Complete
2	Record the version numbers of all test scripts used to perform this qualification test in the table in Section 5.2.2.	Complete/ Not Complete
3	Record the version numbers of the FSW constituents and the Candidate Release on which this test is performed in Section 5.2.3.	Complete/ Not Complete

The following signatures confirm that, using the procedure described in the previous table, the Test Engineer and Quality Assurance Engineer have verified that all versions of test support software, test scripts, and FSW constituents match those identified in Section 5.2.

Date

Time

Test Engineer

QAE

6. TEST PROCEDURE FOR EVTFIL_002

6.1 Test Objective

This test verifies that:

- The output rate of the filtered events is compatible with the spacecraft interface rate of 40 Mbps
- The filter can be configured so that the event stream does not exceed the spacecraft capacity of 104 Gb in any 24 hour period. This test assumes that the actual data volume management policy will be left to the ground operations, and only verifies that the flight software provides sufficient support for implementing such a policy.
- The trigger keeps events consistent with the LAT science objectives. These objectives are specified in the LAT Performance Specification document LAT-SS-00010-1. Objectives relevant to this test are:
 - The effective area for 20 MeV photons at normal incidence should exceed 300 cm²
 - The effective area for 100 MeV photons at normal incidence should exceed 3000 cm²
 - The effective area for 300 GeV photons at normal incidence should exceed 6400 cm²
 - The peak effective area at optimal energy should exceed 8000 cm²
 - The LAT field of view should exceed 2 steradians

The test runs several sets of simulated science data through the Front End Simulator and monitors the FSW performance while the simulations are running.

This Test Objective is broken down into the following Test Sub-Objective(s).

Number	Test Sub-Objective
1	Using simulated photon data with different energies, the test confirms that the fraction of photon events passed by the filter satisfies the effective area and the field of view requirements.
2	The test uses the peak rate and size of passed background events at the orbit interval with high expected background to verify the interface rate requirement.
3	The test verifies that the filter can be reconfigured to reduce the rate of passed events in order to comply with the data volume requirement. Two alternative configuration are used: with tightened event selection criteria and with filter configured to reject all events.

6.2 Test Input Files

The following table identifies all auxiliary files (e.g., Front End Simulator data files, GLEAM data files) used as inputs to this qualification test. Note that not all qualification tests use input data of this type. The “Input File Version Number” column identifies the version number of the auxiliary file being used; alternatively, this column identifies the hardcopy attachment to this document that records the version of the file being used (e.g., “Attachment 1”). The “Path to Attachment” column specifies where in the test repository an electronic copy of the hardcopy attachment has been saved .

Input File	Description of Input File	Input File Version Number (or Specify Attachmnt Number)	Path to Attachment (If Applicable)
orbit_15min.txt	Simulated orbit file for high background orbit		
hibg6min_v7r3p4	FES file simulating cosmic ray background data on high background orbit		
gammaNormal20MeV_v7r3p4	FES file simulating 20 MeV photons		
gammaNormal100MeV_v7r3p4	FES file simulating 100 MeV photons		
gammaNormal300000MeV_v7r3p4	FES file simulating 300 GeV photons		
gammaNormalOneOverE_v7r3p4	FES file simulating 1/E photon spectrum at normal incidence		
gammaAlsoOneOverE_v7r3p4	FES file simulating 1/E photon spectrum with isotropic flux from the zenith hemisphere		
libgfc_db_normal.o	Filter configuration file with default event selection criteria		
libgfc_db_fswtest2.o	Filter configuration file with tightened event selection criteria		
libgfc_db_fswtest3.o	Filter configuration for rejecting all events		
latc_daq.xml	LATC configuration file		
latc_ignore.lat	LATC ignore map		

6.3 Test Output Files

The following table identifies the log files, data dump files, and other output files generated by the test scripts executed during this qualification test. The “Output File Attachment Number” column identifies which attachment to this document is the hardcopy record of this script output data. The “Path to Attachment” column specifies where in the test repository an electronic copy of the hardcopy attachment has been saved.

Output File	Description of Output File	Output File Attachment Number	Path to Attachment
gamma_20MeV.lsf	Science data telemetry, LSF format file for the 20 MeV photon sample		
gamma_20MeV.txt	Text file which contains FES and VSC configuration information for the testbed run with the 20 MeV photon sample		
gamma_100MeV.lsf	Science data telemetry, LSF format file for the 100 MeV photon sample		
gamma_100MeV.txt	Text file which contains FES and VSC configuration information for the testbed run with the 100 MeV photon sample		
gamma_300000MeV.lsf	Science data telemetry, LSF format file for the 300 GeV photon sample		
gamma_300000MeV.txt	Text file which contains FES and VSC configuration information for the testbed run with the 300 GeV photon sample		
gamma_normal_onee.lsf	Science data telemetry, LSF format file for the 1/E normal incidence photon sample		
gamma_normal_onee.txt	Text file which contains FES and VSC configuration information for the testbed run with the 1/E normal incidence photon sample		
gamma_iso_onee.lsf	Science data telemetry, LSF format file for the 1/E isotropic incidence photon sample		
gamma_iso_onee.txt	Text file which contains FES and VSC configuration information for the testbed run with the 1/E isotropic incidence photon sample		
bg_default.shs	Science data telemetry statistics file for the background run with the standard filter configuration		
bg_default.txt	Text file which contains FES and VSC configuration information for the background run with the standard filter configuration		
bg_tight.shs	Science data telemetry statistics file for the background run with tightened event selection criteria		
bg_tight.txt	Text file which contains FES and VSC configuration information for the background run with tightened event selection criteria		
bg_rejectall.shs	Science data telemetry statistics file for the run in which the filter is configured to reject all events		
bg_rejectall.txt	Text file which contains FES and VSC configuration information for the testbed		

Output File	Description of Output File	Output File Attachment Number	Path to Attachment
	run in which the filter is configured to reject all events		

6.4 Test Preparation

After the hardware and software setup has been validated, steps may be required to place the hardware and FSW in an operational mode in which the qualification test can be performed or otherwise complete preparations for the test to begin.

The Test Engineer carries out the following procedure to prepare for qualification tested and records completion of the test preparation steps in the space provided.

Step No.	Description of Step	Step Outcome
1	Confirm that the VSC is powered up: -- The VSC GPS module should display time since last reset.	Complete/ Not Complete
2	Confirm that the Front End Simulator is powered up: -- The power LED on all FES computers (lat-ent1, lat-ent2, ..., lat-ent11) should be on. -- The voltage indicator on the Xantrex power supply marked "FES-FEED" should display $48 \pm 0.5V$, and the current should be above 5A. -- The state indicators on all FES electronic boards should be flashing.	Complete/ Not Complete
3	Confirm that the SIU and the GASU are powered up: -- The voltage indicator on the SIU/GASU Xantrex power supply should display $28 \pm 0.2V$. -- The SIU feed "POWER ON" switch on the regulated feeds bus protection unit (BPU) should be flipped "ON". The SIU voltage indicator on the BPU should display $28 \pm 0.2V$. The SIU current as shown by the BPU indicator should be above 0.4A -- The DAQ feed "POWER ON" switch on the BPU should be flipped "ON". The DAQ voltage indicator on the BPU should display $28 \pm 0.2V$.	Complete/ Not Complete

The Test Engineer and Quality Assurance Engineer verify that all test preparation steps are complete.

Date

Time

Test Engineer

QAE

6.5 Test Procedure

After the test setup has been validated, and test preparation steps are complete, the Test Engineer proceeds with the qualification test procedure itself, as described below, and records the outcome of each step during test execution. The outcome of each step is either "Complete" or "Not Complete"

(for steps which involve no analysis or verification); or, “Pass” or “Fail” (for steps involved in verifying completion of test objectives and sub-objectives).

Step No.	Description of Step	Step Outcome
1	<p>Verify presence of the simulated science data on the FES nodes. On a Linux host, run the command <code>fesdata show '*_v7r3p4'</code></p> <p>It should display</p> <p>Directory :</p> <pre>83503046 23-Dec-2005 07:31:04 ffffffff gammaIsoOneOverE_v7r3p4 7419975 22-Dec-2005 22:20:04 ffffffff gammaNormal100MeV_v7r3p4 3535505 22-Dec-2005 22:15:18 ffffffff gammaNormal20MeV_v7r3p4 15984799 22-Dec-2005 22:31:40 ffffffff gammaNormal300000MeV_v7r3p4 23906380 23-Dec-2005 07:25:12 ffffffff gammaNormalOneOverE_v7r3p4 162501 29-Dec-2005 21:52:04 ffffffff gammaShort_v7r3p4 1955625085 23-Dec-2005 10:32:48 ffffffff hibg6min_v7r3p4 6749999 01-Jan-2006 22:17:56 ffffffff shortIsoOneOverE_v7r3p4</pre>	<p>Complete / Not Complete</p>
2	<p>Launch the data acquisition script. The expected run time for this script is 90 min:</p> <p><code>ltx run EVTFIL_002 --ips=lat-hobbit2,lat-elf21,lat-ent1,lat-elf23,lat-elf24,lat-elf25 --prot=ethernet,xyplex,xyplex,xyplex,xyplex,xyplex</code></p> <p>The data acquisition script performs eight testbed runs using different Monte Carlo event samples and onboard filter configurations and generates sixteen files listed in the table in Section 6.3. The following sequence of operations is performed for each testbed run:</p> <ol style="list-style-type: none"> 1) A set of spacecraft timestamped coordinates and attitudes is scheduled for simulation on the VSC using the VSC interface M7 (“magic seven”) class. The simulated orbit file used for “magic seven” telecommand scheduling is the same as the orbit file used to produce the background simulation. The orbit file format is described at http://confluence.slac.stanford.edu/display/DC2/Orbit+and+Pointing+History 2) The physics run is configured using two or more LPACONFIGURE telecommands. 3) The LPASTART telecommand is sent to the spacecraft. This changes the LAT Instrument Manager operation mode from QUIESCENT to PHYSICS and initiates data taking by the LAT. 4) The LIM “State” telemetry packets are received. The LPA task state, as identified by the LIMTLPASTATE field, is required to be “Running”. The LIM operation mode, as identified by the LIMTOPMODE field, is required to be PHYSICS. The "CmdConfirm" command validation telemetry packet with successful status is expected after the LIM “State” packet with the right properties. The timeout for “CmdConfirm” telemetry reception after the LPASTART telecommand is set to 30 seconds. 5) The FES run is started by talking to the FES system via the LTX interface. FES timing is determined by the simulation for the background runs, and set to 50 Hz for the photon runs. 6) After a waiting period, the FES run is stopped. The length of the waiting period is set to 5 min for each run. 7) The LPASTOP telecommand is sent to the spacecraft. 8) The "CmdConfirm" command validation telemetry packet is expected. After receiving “CmdConfirm”, one or more LIM “State” telemetry packets is received, until the LIMTLPASTATE field of the LIM “State” packet indicates that that the state of the LPA task is “Idle” and the LIMTOPMODE field indicates that the instrument operation mode is QUIESCENT. The timeout for reception of LIM “State” telemetry with right properties is set to 30 seconds after the LPASTOP telecommand. 	<p>Complete / Not Complete</p>

Step No.	Description of Step	Step Outcome
	<p>During each background run the total size of SSR packets and peak rate is accumulated by the science handler on the VSC proxy. During the photon runs all packets received by the proxy on the VSC science stream are written to an LSF file on a hard disk.</p> <p>Upon successful completion of the data acquisition stage, the script displays DAQ script finished YYYY-MM-DD HH:MM:SS.UUUUUU</p>	
3	<p>The data acquisition script automatically runs the data analysis script when all testbed runs are complete.</p> <p>For the fixed energy photon samples, the data analysis script determines the effective area, A_{eff}, as $A_{eff} = N_{pass}/N_{gen} A_{flux}$ where N_{pass} is the number of photon events which passed the on-board filter and were delivered on the VSC science stream N_{gen} is the number of photon events originally generated in Monte Carlo for this energy (before trigger) A_{flux} is the area over which the photon flux was generated (6 m² for all Monte Carlo event samples)</p> <p>The peak effective area is determined from the 1/E normal incidence photon sample. A sliding window with width $\log_{10}(2)$ is used in the $\log_{10}(E / 1 \text{ MeV})$ variable. The number of generated MC events inside the sliding window is assumed to be the same for each window position and determined as $N_{gen} = N_{all} (\text{window width}) / \log_{10}(E_{max}/E_{min})$ where E_{min} is the minimum photon energy in the simulation (20 MeV) E_{max} is the maximum photon energy in the simulation (300 GeV) N_{all} is the total number of events in the sample (before trigger) Peak effective area, A_{peak}, is defined as the maximum A_{eff} over all sliding window positions inside the energy range covered by the simulation. The energy at which the maximum A_{eff} is achieved is E_{peak}. The fraction of photon events passed by the filter at the optimal window position is ϵ_{peak}.</p> <p>The LAT field of view is determined from the 1/E photon sample whose incidence angle is isotropic in the upper hemisphere (2π steradians). The sliding energy window with the same width centered on E_{peak} is used to select the photon log-energy interval for which the field of view is determined. The field of view is calculated as $2\pi\epsilon/\epsilon_{peak}$ where ϵ is the fraction of events passed by the filter for the isotropic sample within the selected energy window.</p> <p>The peak spacecraft interface rate is determined as the maximum size of the data transferred over the VSC science interface each second in the testbed run with the background sample and default filter configuration.</p> <p>The spacecraft interface volume is determined as the average spacecraft interface rate in the background runs multiplied by 86,400 seconds.</p>	<p>Complete/ Not Complete</p>
4	<p>Review the outputs of the data analysis script on the test terminal. The test passes if every line of the output contains the word "PASS". Successful completion of this step fulfills all test objectives and sub-objectives. Example data analysis script output is shown below:</p> <p>Effective area at 20 MeV : 4544 cm² : PASS Effective area at 100 MeV : 8825 cm² : PASS</p>	<p>Pass/ Fail</p>

Step No.	Description of Step	Step Outcome
	Effective area at 300 GeV : 7798 cm² : PASS Peak effective area : 10344 cm² : PASS Field of view : 4.80 sr : PASS Peak SC interface rate : 2.007 Mbps : PASS SC interface volume with default configuration : 96.39 Gb/day : PASS SC interface volume with tightened even selection : 55.89 Gb/day : PASS SC interface volume with "reject all" configuration : 2.328 Gb/day : PASS Combined test status : PASS	
5	Press "Enter" in the terminal to end the test	Complete/ Not Complete

Initial to confirm.

_____ Date _____ Time _____ Test Engineer _____ QAE

7.TEST POST-CONDITIONS AND OVERALL OUTCOME

7.1Test Post-Conditions

The following post-conditions are analyzed and verified by the test script:

No.	Post-Condition	Post-Condition Met? (Yes/No)
1	Upon successful completion of the data acquisition stage, LAT flight software is in BOOT mode	

The Test Engineer and Quality Assurance Engineer verify that all test post-conditions are met.

_____ Date _____ Time _____ Test Engineer _____ QAE

7.2Overall Outcome of EVTFIL_002

Based on the analysis of the test results, the overall outcome of Test EVTFIL_002 is as follows:

- Passed** - all of the expected outcomes for the test were confirmed
- Failed** - one or more of the test outcomes were not confirmed

_____ Date _____ Test Engineer _____ QAE

8.CERTIFICATION

I certify that the information obtained under this test procedure is as represented and the information recorded in this document is complete and correct. Any deviations from test procedures described herein are identified in Appendix A.

Date Test Engineer (Print Name) Test Engineer (Signature)

I certify that the information obtained through execution of this test procedure is as represented and the information recorded in this document is complete and correct. Execution of the test, storage of the results, and verification of outcomes were carried out in accordance with quality standards defined in the GLAST Quality Manual (LAT-MD-00091).

Date Software QA Engineer (Print Name) Software QA Engineer (Sign)

I certify that the information obtained under this test procedure is as represented and the information recorded in this document is complete and correct. The test procedure, as designed and executed, does indeed verify that the FSW functionality under test satisfies the corresponding requirements from the Flight Software Specification – Level III.

Date FSW Manager (Print Name) FSW Manager (Signature)

APPENDIX A: DEVIATIONS FROM THE QUALIFICATION TEST PROCEDURE

This section details any deviations from the hardware configuration, software configuration, or test procedure followed during the execution of the test or tests described in this Qualification Test Procedure document. All deviations from the approved procedure are agreed to by the Test Engineer and the Software Quality Engineer during the test execution session. All deviations must be reported during the Post Qualification Test Review, where their impact on the test results will be evaluated.

Hardware Deviations

Describe any deviations from the hardware configuration defined in Section 5.1. Name the hardware that was modified and describe the modifications. If hardware is *replaced* during execution of the test, name the replaced hardware, the manufacturer, and list an identification number (e.g., GLAT ID number).

Software Deviations

Describe any changes made to the software configuration under test or the software configuration used to support test execution, as defined in Section 5.2. Give version numbers of all FSW packages and test packages that were modified. Describe how the contents of the modified software load were verified. Describe these deviations for each test that was modified.

Procedural Deviations

Specify any deviations from the test procedure for the test being executed. If this document contains more than one test procedure, list the procedure by number (e.g., “MEMMGT_001”). List by number the steps modified or skipped. Provide a numbered sequence listing any added steps. Describe these deviations for each test that was modified.
